



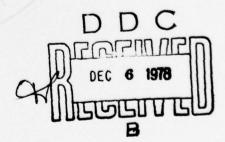
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BIAXIAL TESTING TECHNIQUES OF THIN-WALLED TUBULAR SPECIMENS

JAMES H. RAINEY, RONALD A. SWANSON, and SHUN-CHIN CHOU BALLISTIC MISSILE DEFENSE MATERIALS PROGRAM OFFICE

September 1978

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ARMY MATERIALS AND MECHANICS RESEARCH CENTER Watertown, Massachusetts 02172

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ABSTRACT

A biaxial testing technique of a thin-walled tubular specimen is described. The specimen was loaded with a combination of axial tension/compression and internal/external pressure. The technique can be used to test tubular specimens of any material. The yield surface of 2014-T651 aluminum was determined to illustrate the testing technique and data analysis.

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I. INTRODUCTION

The response of materials under multiaxial stress states beyond the elastic range is an essential part of the mechanical properties required by designers of structures. This report describes a technique to test material in the form of a thin-walled tube under axial and tangential (hoop) stresses. A brief description of the mechanical testing machine and information of an automated data acquisition and control system are presented. The automated control parameter determination and some experimental data of aluminum alloy 2014-T651 are also discussed.

II. TESTING TECHNIQUES

The equipment (Figure 1) required to perform the biaxial testing of tubular specimens can be described in three major components: namely, the medium strain rate machine, intensifier system, and data acquisition and control system.

The medium strain rate machine (MSRM) is used to generate the axial tensile or compression stress, while the intensifier system provides the tangential (or hoop) tensile or compression stress in a tubular specimen. The data acquisition and control system records and stores all data and generates command signals for both machines.

A. Medium Strain Rate Machine

The MSRM is a dual-mode test machine capable of generating 140,000 pounds axial tension or compression. The two modes of operation are an open loop system, and a closed loop servohydraulic system. The open loop system has the capability of strain rates from 1 to 50 $\,\mathrm{sec}^{-1}$, but it is not normally used in biaxial testing. The closed loop system will produce strain rates from 10^{-5} to 10^{-1} $\,\mathrm{sec}^{-1}$.



Figure 1. Automated materials characterization system.

The MSRM control panel gives the operator a selection of four different feedback control modes: load, displacement, strain, and optional. Therefore, with proper command inputs, tests at constant rates of load, displacement, and strain are performed. The operator may select one of many different load cells to achieve the best control over the desired testing range. The machine is equipped with a 15-gpm hydraulic power supply and a 15-gpm servovalve. Various fail-safe and limiting devices which either display a warning light or cause a machine shutdown are incorporated into the control system.

The MSRM load frame is designed for a stiffness greater than 15×10^6 lb/in. and has a total machine stretch of 0.005 inch at 140,000 lb load.

B. Intensifier System

The closed loop servohydraulic intensifier system used for the biaxial internal and external pressure testing is capable of generating a pressure of 100,000 psi. The intensifier vessel has a 1-inch inside diameter with an 8-inch length giving a volume of about 6 cubic inches. A Haskell air-driven pump is used to fill the intensifier system with Stoddard solvent as the pressurizing fluid. This fluid has a freezing point well in excess of seven kilobars.

The intensifier control panel gives the operator a selection of four different feedback loops: load, strain, displacement, and optional. With the proper command input, constant rates are possible under closed loop control. In essence, the intensifier system is a testing machine by itself. Different pressure transducers can be used so that the best control is available over the pressure range of interest. Pressure control is used in the optional mode. System pressure is also monitored using a precision Heise gage with a capacity of 100,000 psi. The intensifier uses the same hydraulic power supply as the medium strain rate machine.

C. Data Acquisition and Control System

The computer configuration consists of a central processor, 4K words of basic memory plus 12K words extended memory, real-time clock, relay register, 1.6 million word disk, two magnetic tape drives, teletype and line printer, display screen, multiplexed analog-to-digital converter (16 channels), and three digital-to-analog converters. The system interface includes eight active filters and scaling amplifiers.

The digital computer has a central processor which uses its memory to hold the operating system, to store programs during execution, and for temporary storage of data.

Command signals are generated by the central processor and sent to the digital-to-analog converters at a predetermined interval by the real-time clock. The converter changes binary numbers (12BITS), which are the internal information base of the computer, to an analog voltage (± 10V) that is acceptable to the servocontroller of the test machines. The controllers operate in a closed-loop mode generating a signal which drives the electrohydraulic servovalves. The servovalves regulate the flow of oil from the hydraulic power supply, to the test machine actuators, which deform the specimen. The specimen load, pressure, strains, and displacements

are continuously monitored and input into the computer. These analog signals are filtered to remove noise and scaled to match the input range of the analog-to-digital converter. The multiplexer selects one of the 16 channels for input to the analog-to-digital converter according to the program. A sample-and-hold circuit holds the value while it is converted to a binary number for input into the central processor. The data are stored in memory, then either displayed on the screen, printed, or stored on magnetic tape or disk. A second command signal may now be sent to other machines through a different digital-to-analog converter channel, based on the data received from the first signal. The real-time clock is used to determine sampling intervals, command intervals, and signal events. The teletype or console is used to input programs and parameters to the central processor. The magnetic tape units and disk provide fast access mass storage for programs and data. The relay register is used for additional control of the test machines and external equipment.

D. Specimen Configuration and Grip Fixtures

The thin-walled biaxial tubular specimen configuration shown in Figure 2 is designed for both internal and external pressure testing. The specimen designated as 601066-02 has a shorter gage length and overall length in comparison with the 601066-01 specimen. The shorter specimen is used to prevent premature buckling failure before the material reaches its yield stress when the specimen is subjected to compression in both axial and tangential directions.

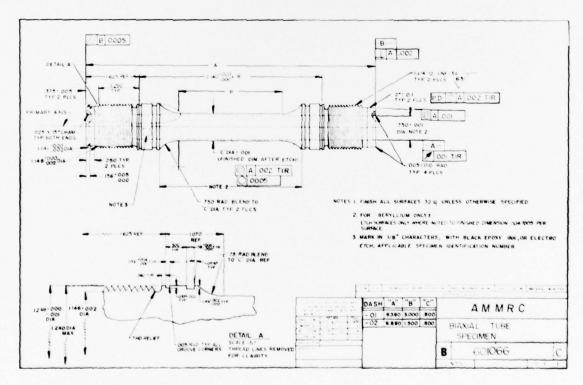


Figure 2. Specimen configuration for internal and external pressure tests.

In order to maintain the stress state in the specimen as close to a state of plane stress as possible, the wall thickness at the gage length is 0.025 inch, which results in an 8% difference in the tangential stress at the inner and outer radii.

The tensile tangential stress is created through the use of an internal plug assembly as shown in Figure 3, while the compression tangential stress is generated with a sleeve arrangement as shown in Figure 4. The external pressure assembly is shown in Figure 5. The pressurized fluid in the plug or sleeve is controlled by the intensifier system, and the pressure inside the plug or sleeve is contained by the use of "O" rings at both ends of the assembly.

The other precaution one must take in testing tubular specimens is that it is essential to maintain a homogeneous strain field in the gage section. This can be accomplished by a precise alignment procedure. A threaded collar containing eight bolt holes is threaded on each specimen end, having approximately 0.25 inch of the specimen exposed. The taper on the exposed ends of the specimen allows an initial alignment of the specimen with respect to the loading frame (see Figure 6). Bolts are then inserted through the collar and threaded to the load frame. While tightening bolts to the load frame, the two axial strain gages on the specimen are monitored to assure that, first, no bending moment is induced through uneven tightening of the bolts; and secondly, zero axial load is maintained. This procedure works very well as test results show that both axial strain gages record nearly identical strains until the specimen approaches failure.

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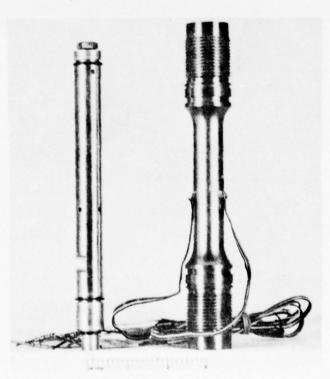


Figure 3. Biaxial internal plug and specimen. 19-066-7/AMC-78

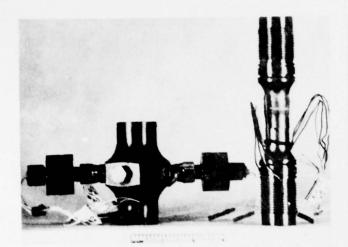


Figure 4. Biaxial external pressure sleeve and specimen. 19-066-6/AMC-78

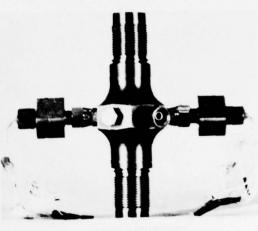


Figure 5. Biaxial external specimen assembly. 19-066-5/AMC-78

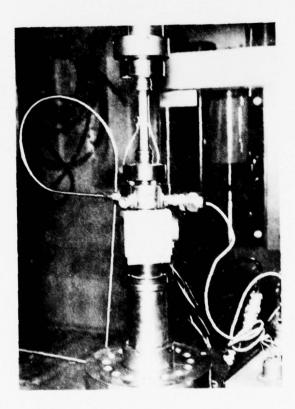


Figure 6. Grip arrangement for biaxial internal pressure test.

E. Specimen Strain Gaging

All biaxial specimens are instrumented with EPOS-062TT-120, 90° "tee" rosette strain gages to measure the axial and tangential strains simultaneously. The gaging area is prepared in accordance with the manufacturer's instructions for the particular adhesive being used. Gages are bonded to the specimen with EPY-350 adhesive using a clamping pressure of 10 to 15 psi, and cured for two hours at 350 F. A protective coating of flowable, room temperature curing silicone rubber is used to protect the strain gages from environmental and handling damage. The detailed procedure for gaging specimens can be found in Micro Measurement Instruction Bulletin No. B-127-3, B-130-3, and B-137-2.

At the beginning of this study, four sets of "tee" rosette strain gages were used. They were located at the center of the gage section and placed 90° from each other along the circumference of the tubular specimen. However, since the alignment procedure described in the last paragraph creates a very good homogeneous strain field, it was decided that two sets of "tee" rosette strain gages are sufficient to measure the strain field in the tubular specimen. Test results reported are average values of the two sets of rosette strain gages located diametrically from each other.

F. Testing Control Parameters

Since the medium strain rate machine and intensifier system have separate servohydraulic mechanisms, the axial and tangential stresses (or strains) can, in principle, be controlled at different rates. This operation requires the use of a more sophisticated control theory because the material deformations are interrelated through the Poisson's effect. In this study, each test was carried out along a proportional load path which provides a constant effective strain rate in terms of either axial or tangential strain rate (see the mathematical formulation in the next section), that is, the ratio of the axial and tangential stresses is a constant throughout a test. As described in the section on "Data Acquisition and Control System," the signals from load cell and strain gages were magnified by the gage conditioning units; therefore, it would be logical to use the dominant strain as the control parameter. If the dominant strain is not controlled, it is possible that machine "runaway" might occur and the specimen would fail prematurely. For the case when the ratio of axial stress to tangential stress $(R=\sigma_A/\sigma_o)$ is greater than or equal to unity, the axial strain is the control parameter and the computer program sends the command signal to the intensifier system to either increase or decrease the internal (or external) pressure so that a constant stress ratio will be maintained. On the other hand, when R<1 the tangential strain will be the control parameter and the axial load is determined through the computer program. This procedure provides the stress ratio within 1% of a desired value.

III. DATA ANALYSIS

The biaxial test data presented here are in the form of effective stress-strain curves to illustrate strain hardening behavior and yield and failure stresses in the two-dimensional stress space. The data are presented in terms of the axial and tangential stresses σ_A and σ_T ; the corresponding strains are ε_A and ε_T .

Before discussing experimental results a definition of yield must be made. The definition used here is based on the square root of the second invariant of the stress deviator $\sqrt{J_2}$ plotted against the square root of the second invariant of the strain deviator $\sqrt{I_2}$, where in this study,

$$\sqrt{J_2} = \frac{1}{\sqrt{6}} \left[\left(\sigma_A - \sigma_T \right)^2 + \left(\sigma_T - \sigma_R \right)^2 + \left(\sigma_R - \sigma_A \right)^2 \right]^{\frac{1}{2}}$$

$$\sqrt{I_2} = \frac{1}{\sqrt{6}} \left[\left(\epsilon_A - \epsilon_T \right)^2 + \left(\epsilon_T - \epsilon_R \right)^2 + \left(\epsilon_R - \epsilon_A \right)^2 \right]^{\frac{1}{2}}$$

and σ_R is the radial stress and ϵ_R the radial strain.

The axial and tangential strains ε_A and ε_T are measured strain gage values on the outside surface of the tubular specimen. The radial strain ε_R through the wall of the tube is calculated using elasticity equations during elastic loading and the assumption of incompressible flow for the plastic components of strain after "yield."

The stresses are calculated from the axial load and the pressure. In the case of a tubular specimen subjected to axial load and internal pressure, the axial stress σ_{A} is

$$\sigma_{A} = \frac{F}{\pi (r_{o}^{2} - r_{i}^{2})}$$

where F is the axial load and r_0 , r_i are the outer and inner radii and $r_0 \ge r \ge r_i$ and the tangential stress σ_T is

$$\sigma_{T} = \frac{P_{i} r_{i}^{2}}{r_{o}^{2} - r_{i}^{2}} \left(1 + \frac{r_{o}^{2}}{r^{2}} \right)$$

where P_i is the internal pressure.

In the case of a specimen subjected to axial load and external pressure, the vertical component of the hydraulic pressure acting on the surface of the fillet at both ends must be taken into consideration when the axial stress is calculated

$$\sigma_{A} = \frac{F}{\pi (r_{o}^{2} - r_{i}^{2})} + \frac{P_{o} (r_{s}^{2} - r_{o}^{2})}{(r_{o}^{2} - r_{i}^{2})}$$

where r_S = shoulder radius of the specimen (Figure 2) and P_O = external pressure. For specimen used in this study r_S = 0.624 inch.

The tangential stress

$$\sigma_{T} = -\frac{P_{o} r_{o}^{2}}{r_{o}^{2} - r_{i}^{2}} \left(1 + \frac{r_{i}^{2}}{r^{2}} \right) .$$

Once the deviatoric stress-strain curves are obtained, yielding is defined as the intercept of a line drawn parallel to the initial linear portion of the curve at 0.2% strain offset.

As it was discussed in the last section, all tests presented here are performed under proportional load conditions. In these tests, one strain rate was controlled to be constant while the second servo was used to maintain a constant stress ratio. Selection as to which direction would be maintained at constant strain rate was usually determined by the dominating stress. It will be shown below that tests performed under these control conditions also have a constant effective (or deviatoric) strain rate. The effective strain rate is defined as

where the superscript P denotes plastic strain, and the dot indicates rate of change.

We further assume the Prandtl-Reuss flow rule

$$d\varepsilon_A^P/S_A = d\varepsilon_T^P/S_T = d\varepsilon_R^P/S_R$$

or

$$\epsilon_{A}^{P}/S_{A} = \epsilon_{T}^{P}/S_{T} = \epsilon_{R}^{P}/S_{R}$$

where S's are deviatoric stress components, and given as follows:

$$S_A = \sigma_A - 1/3 (\sigma_A + \sigma_T + \sigma_R) = 1/3 (2\sigma_A - \sigma_T)$$

$$S_T = \sigma_T - 1/3 (\sigma_A + \sigma_T + \sigma_R) = 1/3 (-\sigma_A + 2\sigma_T)$$

$$S_R = -1/3 (\sigma_A + \sigma_T)$$

with $\sigma_R = 0$.

The flow rule then becomes

$$\frac{\stackrel{\bullet}{\varepsilon_{\rm A}^{\rm P}}}{1/3(2\sigma_{\Lambda} - \sigma_{\rm T})} = \frac{\stackrel{\bullet}{\varepsilon_{\rm T}^{\rm P}}}{1/3(-\sigma_{\Lambda} + 2\sigma_{\rm T})} = \frac{\stackrel{\bullet}{\varepsilon_{\rm R}^{\rm P}}}{-1/3(\sigma_{\Lambda} + \sigma_{\rm T})}$$

If we define the stress ratio, $\beta = \sigma_T/\sigma_A$, we have

$$\frac{\dot{\epsilon}_{\mathrm{T}}^{\mathrm{P}}}{\dot{\epsilon}_{\mathrm{A}}^{\mathrm{P}}} = \frac{2\sigma_{\mathrm{T}}^{-\mathrm{O}}A}{2\sigma_{\mathrm{A}}^{-\mathrm{O}}\mathrm{T}} = \frac{2\beta-1}{2-\beta} \equiv \alpha$$

which gives $\dot{\epsilon}_T^P = \alpha \dot{\epsilon}_A^P$.

The assumption of incompressibility provides

$$\dot{\varepsilon}_{A}^{P} + \dot{\varepsilon}_{T}^{P} + \dot{\varepsilon}_{R}^{P} = 0$$

and

$$\frac{\cdot}{\varepsilon_R^p} = -(1+\alpha) \frac{\cdot}{\varepsilon_A^p}$$
 .

Then substitution into the definition of effective strain rate yields

$$\frac{e^{P}}{eff} = \frac{e^{P}}{e_{A}} (1+\alpha+\alpha^{2})^{\frac{1}{2}}$$
.

Thus, for a selected proportional load path $\beta = \sigma_T/\sigma_A$ and a prescribed constant strain rate $\dot{\epsilon}_A^P$, the effective strain rate can be determined.

Two computer programs were written for the purpose of testing tubular specimens under biaxial stress states. The program "BIAX" was written to automatically record the axial load from the load cell, the pressure from the pressure transducer, and the axial and tangential strain values from the two sets of strain gages on the specimen. These values were stored on a magnetic tape through analog-to-digital converters. The program "BIANDV" was written to calculate the deviatoric stress, strain, and strain rate from the data recorded on the magnetic tape, and the results were printed on the line printer. A plotting routine was also included in "BIANDV," so the deviatoric stress-strain curve was automatically plotted on an X-Y recorder for each test.

The listing of these two programs are given in the Appendix.

IV. EXPERIMENTAL RESULTS AND DISCUSSIONS

The described biaxial testing technique was used to determine the yield surface of aluminum alloy 2014-T651 on the axial and tangential stress (σ_A, σ_T) plane. The tubular specimens were machined from 2- and 3-inch-thick rolled plates of 2014-T651 aluminum. Yield surface is shown in Figure 7. In the first quadrant (i.e., $\sigma_T > 0$, $\sigma_A > 0$) some specimens failed before reaching the 0.2% offset, but in the plastic region; in these cases, yield was defined as the maximum stresses reached during the test. The experimental data were also fitted to a quadratic equation with the least-square method

$$A\sigma_{\Lambda}^2 + B\sigma_{\Lambda}\sigma_{T} + C\sigma_{T}^2 + D\sigma_{\Lambda} + E\sigma_{T} = k^2$$

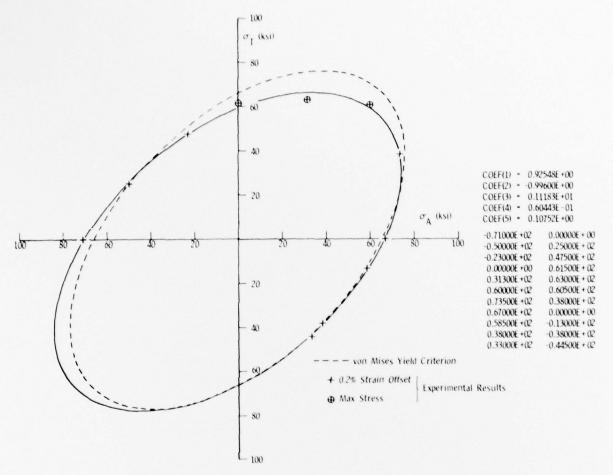


Figure 7. Yield surface for 2014-T651 aluminum alloy.

where k is the yield stress, 67 ksi, in simple tension. The coefficients are given as follows:

A = 0.925

B = -0.996

C = 1.118

D = 0.060

 $E \approx 0.107$

which indicates that the behavior of the material agrees reasonably well with the von Mises yield criteria. Furthermore, the experimental results presented in this report were compared with results obtained by other investigators, e.g., Reference 1, and they agree very well. This comparison serves as a verification of the testing techniques and data analysis procedure presented here.

 REID, R., J., JONES, A. H., and GREEN, S. J. Characterization of 2014-T651 Aluminum Alloy. Terra Tek, Salt Lake City, Utah, Contract DAAG46-74-C-0019, Final Report, AMMRC CTR 74-68, November 1974.

APPENDIX. BIAX AND BIANDV PROGRAMS

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                 STRAIN CONTROL ON AXIAL DIRECTION
                  PRESSURE CUNTROL ON THETA DIRECTION
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0067
0078
                HEAD (4, 2011) US
                RS=05/2.
0071
         C
                SHOULDER ANEA
         C
         C
2072
               45=3.14154+(K5++2-H0++2)
de75
               CUNE =PCAL +AS/511.
         1
                STHAIN GAGE INFORMATION
         C
```

```
0074
            25 00 4V1 1=1.2
0075
                WHITE (4, 1075) I
0076
          1073 FORMAT(" #", IZ," HES. # ",5)
0077
                READ(4,2001) HG
0100
                WRITE (4, 1072) I
          1072 FURMAT(" A", 12, " 6.F. & ", 5)
0101
5016
                HEAU(4,2001) GF
0105
                WHITE (4, 1074) I
          1014 FORMAT( " 4", 12. " SHUNT . ", 51
0104
0105
                REAU(4, 2001) HS
1100
                5TC(1)=(1+K5/(K5+KG))/GF
0107
           401 CUNTINLE
         1
                  CUMPLITATIONS FOR EFFELTIVE STRAIN HATE
                  AND TEST TIME
         C
€11€
                IF (HATIO. GE. 1) BET=1./HATIC
                IF (HATIU.LT.1) BETHATIU
V111
0112
                IF (AXIAL. NE. T. AND. INT. EU. Y) BETA-BET
                IF (AXIAL. EG. T. AND. INT. NE. Y) HET = - HET
2115
3114
                ARITE (4, 1070)
          1010 FURMAT( + EFFECTIVE STHALL HATE . .. 5)
2115
2110
                MEAU(4,2001) EFF
                ALP& (2. HET-1)/(2. - HET)
0111
0150
                EUDT=EFF/SUNT(1.+ALF+ALF++2)
1510
                WHITE (4, 1, 71) EUCT
          10/1 FURMAT( * CONTROLLED S.H. = 1,F12.51
IF(HATIO.OF.1) TIME=STC(1)/EDGT
5515
6153
                IF (HATIU.LT.1) TIME = STC(2) / EOCT
0124
0125
                WRITE (4, 1200) TIME
            11 ANITE (4, 1021)
0120
         C
                 HIT HETURK KEY TO START TEST
         C
2121
                TEAD (4,2000) LONT
6150
                INDA1=1
                IF (AXIAL. NE.T) INUALE-1
2151
0132
                IF (AXIAL.NE.T) CALL DIGA(1.+)
2155
                HEHL
£134
                A465.14154*(NU**2-H1**2)
                T[=(H]++2+(HD++2++++2))/(N++2+(HC++2+N[++2])
0135
                IF (INI, NE, Y) TT= (HD++2+(R1++2+R++2))/(R++2+(R(++2+R1++2))
2150
                CUNV=2048+XLCAL/(AA+TT+PCAL+511+HAT10)
V137
0140
                NUAADSH
V141
                NP15=250
7142
                CALL CLRPLT(200, PLTHUF)
0143
                KKSTIME/20.+1
V144
                NUASZOUN
1145
                NIPIS= (KK+26481+8
                CALL REALTM (DATBUF, 50, 8, 6, NTFTS)
1140
2147
                CH=(KK+2048)/TIME
         C
                INITIALIZE DATA TO ZENO
1150
               00 300 1:1.256
```

10

```
A(I) ....
0151
0152
                H(I) . U. U
                C(1)=0.0
0155
                0(1) ...
0154
0155
                t(1) ....
                F(1) =0.0
2150
0157
          300 CALL PLUT(1..65,.5)
                CALL CLUCK ( P. CH)
W160
2161
                00 301 1 =1,256
                IF (1. EG. 256) NOA4U#7
6105
2163
                UAAUA, IEL SNE GO
                CALL 584 (5, 1885)
6104
                IF (ISNS5.EQ. 1) GC TO 140
2165
                N. SEAAA
Pine
0167
                0.3033
                00 505 N#1, NK
0170
V171
                CALL SSA(C. ISNSU)
                IF (15NSU.EG.1) GO TO 334
2172
0175
                IF ( N. NE . KA) GU TO 364
                 MANP ON CONTROL GALE
                IF (MATID.LT.1) DO TO SAN
217-
0175
                DAI . DAI + I NOAY
                CALL UTUA(1,041)
GU TO 304
0176
6177
                1+5404540
6500
           580
1650
                CALL DTOA(2,DA2)
         C
                 SAMPLINE HUUTINE
         C
            504 4(1) = ACC(x)
2333
0203
                8(1) = AUB(X)
                C(I) = ADB(A)
0204
0205
                D(I) . ADH(X)
                E(1) = ADd(x)
0200
2207
                F(1) = ADO(x)
                IF (INT.EQ.Y) C(1) =-C(I)
delk
1159
                IF (IN1.EQ.Y) D(I) =-D(I)
                IF (INT.NE.Y) A(I) = (A(I) + XLCAL/511. + AH5(C(I)) + CONE) +511./ XLCAL
0216
1213
                AAA = AAA + A ( I )
4214
                CC . CC + C(I)
4215
            303 CUNTINUE
         C
                 AVENAGE DATA HEFORE STORAGE
         C
0210
                A(1) = 444/AK
0217
                C(I) = CC/KK
1550
                IF (MATIU. Ut. 1.0) 60 TO 3A3
         C
                 COMMAND TO MSRM
         C
         C
1223
                SIGZ=(XLCAL/(AA))
                SIGTEAHS(C(1) . PCAL . TT) . NATTO
3555
                DA1=5161/5162.4.
6550
```

```
IF (AXIAL. N. T) DA1 = - DA1
4550
0225
                IF (INT. NE. Y) HAI + DAI + C(I) + CONE + 511./ XLCAL + 4.
4626
                IF (ABS (JA1) . 67 . 2047) 60 70 400
1550
                CALL SSM(U. ISNSU)
2230
                IF (15NSd. EG. 1) GC TL 313
               CALL UTUA(1, MAI)
2231
                GU TO 313
8232
        C
                 COMMAND TO INTENSIFIER
0235
           303 0A2 . A0S (A(I) . CONV)
0234
                IF (DAE. GT. 2047) GO TO 900
0235
                CALL SSM(U, ISNSU)
0230
                IF (1885), to. 1) 60 TU 515
                CALL DTOA(2,0A2)
0237
           313 CUNTINUE
0240
8241
          SHE CUNTINUE
         C
                 PLOT AXIAL VS THETA STRESS ON SCUPE
         C
1242
                CALL PLUTH(1,A(1)+1.5/1/24,+.65,C(1)/1/24,+.5,1)
               CUNTINUE
6450
          501
2500
                GU TO 100
               CUNTINUE
6502
          8 V V
2240
                WHITE (4, 1011)
                GO TO 160
0247
0250
          400
                WHITE (4,1227)
                CUNTINUE
0251
          120
                 UUMP DATA TO TAPE
         C
         C
2650
                WHITE(4, 1016)
                L = 1
0253
1250
                MHITE(1"L)A
                F=5
1255
0256
                WHITE (1'L) D
0251
                L=3
1200
                WHITE (1'L) C
1959
                L=4
2053
                WHITE (1"L) U
1265
                L=5
Victo
                WHITE (1'L) t
1'005
                L=0
                HHITE (1"L) F
1200
                WHITE (4, 10Ha)
1950
2270
          1088 FURMAT(" NETURN TO END PHOG. & REMOVE SUFTRARE CLAMP")
                90 (1885, P) GE
0271
                CALL HCPEN(d)
5155
                CALL MCPEN(2)
2213
2214
          103 CALL EXIT
```

V

```
OS/6 FORTRAN IN 3.05
                             ANALYSIS OF & CH OF HIAX
           PRUGNAM HIANUV
        C
            HATIO LESS THAN 1.
        C
        C
                STRAIN CUNTRUL ON AXIAL DIRECTION
        E
                STRAIN CONTROL ON THETA DIRECTION
        C
        C
                FIGURES LOAD CUMMAND FROM PRESSURE REALING
        C
        C
            HATIO GREATER THAN 1.
        C
                STHAIR CUNTRUL ON AXIAL DIRECTION
        C
        C
        C
                PRESSURE CUNTRUL ON THETA DIRECTION
        C
                FIGURES PRESSUR COMMAND FROM LUAD HEADING
        C
        C
        -
               CUMPUTES DEVIATORIC STRESS-STRAIN CURVE
        C
        c
               PLOTS DEVIATORIC STRESS-STRAIN CLAVE
        C
        C
        C
               USES SUBROUTINE XYMEC
        C
               OIMENSIUN 4(256),0(256),0(256),0(256),E(256),F(256)
6665
0005
               DIMENSION HS(4), HG(4), STC(4)
              COMMON/G/A, H, C, D, E, F
4690
0005
               DEFINE FILE 1(0,250,U,L)
               Yalky
0469
        C
        C
                HEAD DATA FROM TAPE
        C
0001
               1 .1
               READ(1'L) A
0010
0011
               1=5
2012
               READ(1'L) H
0013
               1. . 3
               REAU(1'L) C
4144
0015
               L=4
               READ(1'L) D
0616
0017
               L=5
               READ(1"L) E
0050
1530
               L=0
               READ(1"L) F
2538
        C
                 CHOSS PLOT DATA ON SCOPE
        C
        C
                 RETURN TO EXIT FROM PLT6
        C
0023
               CALL PLTS
4654
         1000 FORMAT(" TEST NUMBER IS ", 244)
         1001 FURMAT( ' UATE IS ",13,13,15,//)
2500
         1002 FURMAT( LUAD CAL - LBS ')
0020
```

4

```
1003 FORMAT (" PHESS. CAL - PS1 ")
1004 FORMAT (" INSIDE DIA. ")
1530
2030
          1005 FORMAT( OUTSIDE UIA. ')
C. 31
2232
          1006 FURNAT( TOTAL TEST TIME (SEC) ")
          1010 FUHMAT( STHESS HATTO ")
2033
          1011 FURMAT( * INTERNAL PRESSURE ? ". . )
0034
          1021 FORMATI TYPE HETURN TO GU ")
0035
          (HAS) TAMBUT SUBS
0030
VV37
          2001 FORMAT(IIO)
          2002 FORMAT(141)
0040
         C
                 INPUT CALIBRATION VALUES
         C
....1
                WRITE (4, 1000)
                HEAD(4, 2000) TEST1, TEST2
6645
                CALL UATE (J1, J2, J3)
0043
                WAITE (4,1001) J1, J2, J3
....
2645
                MHITE (4, 11 02)
                HEAD(4, 2011) XLEAL
. 40
9847
                WHITE (4, 1005)
                HEAD (4, 2001) PCAL
. . 5.
                WHITE (4.1:14)
2251
66.25
                ME 40 (4,2011) UI
2.55
                .5/1de18
1654
                **116 (4,1,05)
2055
                DU (1.65,4) UA3H
415e
                ML . UC/2.
2057
                HENL
                ##11E(4,1011)
******
                REAU(4, 2002) INT
....1
3000
                amite (4, tell)
                ME40 (4,2001) MATIO
1103
                ANTIE (4, 1, 100)
....
         C
                INPUT STRAIN GAGE INFORMATION
         C
         C
0205
                MEAL (4,2401) TIME
2660
                mm17t(4,1101)
0267
                MEAU (4,2001) GF
          1101 FORMAT( " GAGE FACTOR ")
2870
0071
                00 12 1:1.4
2072
                WHITE (4, 110c) I
          1102 FURMAT(" .", IZ, " GAGE RES. ")
0073
                NEAL (4,24.1) HG(1)
. . 74
0.75
                00 13 141,4
2676
                WHITE(#,11/3) I
2211
          1123 FORMAT(" &", IZ, " SHUNT HES, ")
2126
          1 3
               READ(4, 2001) RS(I)
2111
                00 14 111,4
010c
          14
                STC([]=(1-NS([])/(NS([])+NG([])))/U++100
         C
         0
                 SETUP FOR DETERMINING PRINCIPLE STRESSES
         C
2175
                AA=(R1++2+PCAL/511)/(RC++2+K1++2)
                IF (INT. NE. Y) A40 (HU. +2 + FCAL /511) / (RU ++2 + 41 ++2)
4104
```

V

```
0105
               $161C=(xLCAL/511.)/13.14149+(HO++2-+1++2)]/1864.
1100
               $162C=44.(1.+(HU..Z/H..Z))/1068.
0107
                I+ (INT. NE. Y) SIG2C+AA+ (1++1++2/R++2)/1248.
2112
                $163C****(1.-(HO**2/H**2))/1000.
.111
               IF (INT. NE. Y) SIGSC = A4 + (1 - (HI + + 2/H + + 2))/1000.
             DETERMINE IF THU AXIAL CH THE THANSVERSE
               GAGES DIFFER BY MURE THAN 5% OF CALIMNATION VALUE
4114
               00 20 1:1,250
               E1 . 0 (1) . STC(1)
2113
2114
               E2.1(1).STC(2)
2115
               E3+E(1)+STC(3)
0115
               E4 = F (1) + STC (4)
2117
               DEA= (ARS(E1) - AHS(E3))
               DET= (AHS(E2)-AHS(E4))
4120
               De1 ... 5 . STC(1) . 511
1516
0155
               DE2=. 15.510(2) +511
                IF (UEA. GT. DE 1) GO TO 21
2123
                1 ( UE1 . UT . DE2) GO TO 22
4515
2125
          20
               CONTINUE
               GU TO 27
0515
               HHITE (4, 1104) I
0127
          21
0130
                60 TO 21
C131
          55
                MHITE (4,1145) I
2132
          1104 FURMAT(" AXIAL GAGES DIFFER THAN MORE THAN SX AT PT 4", 13)
1135
            27 WHITE (4,1141)
2134
          1141 FORMAT( MOUULUS X10 .. 6 PS1 "1
0135
                MEAL (4, 2011) EMOO
:130
                WHITE (4, 1142)
          1142 FURNAT( POISSONS HATTO . ')
0137
                REAL (4, 2411) U
2146
2141
                EMOUSEMOD . I . . . . . . . . .
          1185 FUNMAT (" THETA GAGES UIFFER MORE THAN SE AT PT A", 151
5415
2145
               J = 1
               00 30 1-1,256
2144
0145
                IF (I.NE. J. OK. ISNSU. EG. 1) GC TC 40
2140
                wwitt(3,5ene)
0147
          SUMP FORMAT (1H1)
0150
                walte (3, tore) TESTI, TEST2
2151
                walte (3,1001) J1, J2, J3
5615
               WHITE (3, 1106)
2153
          1106 FORMATC
                          AXIAL
                                   THETA
                                            RADIAL ANTAL
                                                              THETA .
                 TIME
              1.
                                                        UEV "1
                           STHAIN
                                      FEV
                                                DEV
1154
               WALTE (3, 1007)
2155
          1897 FORMATE
                          STHESS STRESS STRAIN
                                                             STRAIN".
              1' SEC
                           RETE
                                    STALSS STRAIN RATE
                                                              ..//1
¥156
               J=J+52
         C
         C
                COMPUTE PHINCIPLE & DEVIATORIC STRESSES
2157
               S1 . S161C . A(1)
2100
               Se . $1620 . C(1)
.161
               $3.$163C.C(1)
               E1=((b(1)+STC(1))+(E(1)+STC(3)))/2./511.
1106
```

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2103
               E2.((0(1).STC(2)).(F(1).STC(4)))/2./511.
2104
               11. (52-53) ...
0105
               72 = (53 - 51) + + 2
2100
               13.(51-52) ...
2107
               $1GU=(1/Sunt(6)) +SUNT(T1+T2+T3)
                COMPUTE NAUIAL STRAIN AND UPVIATORIC STRAIN
2170
               £(1) . $160
2171
               E3.1/E~00.($3.1000.-U.($1.52).1000.).100.
2172
               F1=((E2-E3)/100.1.02
2173
               F2.((E3-E1)/100.) ...
2174
               F3=((E1-E2)/100.) ...
2175
               ELD. (1/5GNT(6)). SURT (F1.F2.F3).100.
2176
               F(I) *EEU
0177
               TM = (TIME /250) + I
               11.11ME/256
0229
               CALL 55+ (0, 15NS.)
1959
               Ir (1.67.15) GU TO 50
2020
2285
               SH.0.0
9564
               WHITE(3,1130) $1,52,53,61,62,1M,5K,5160,660
2205
               GO 10 30
0220
               IF (MATIG.LT.1) SH+(U(19-0(1-15))+STL(2)/(T1+15)/164./511.
0207
               IF (HATIO, GE.1) SH= (B(1)-B(1-15)) +STC(1)/(T1+15)/102./511.
0212
                IF (MATIO.LT.1) 60 TO 37
          1130 FURNAT (3F8.2,2F8.3, F8.2, F9.5,2F8.2, F4.5)
1155
2216
            37 EUNIINLE
0213
               IF (52.EU.U.W) GC TO 888
                DETERMINE EFFECTIVE STRAIN HATE
         C
4159
               HAT . 51/52
               484 IF (52. EU. U. U) HAT - 51 - 49494
2215
               IF (MATIO.GE.1) BET#1./MATIC
0150
               IF (HATIU.LT.1) HETERATIC
2217
               IF (NAT, LT, 0) BETR-BET
3550
               ALP . (2. . BET - 1.) / (2. - HET)
1555
               DEV3.58.SURT(1. - ALP-ALP. - 2)
2225
               IF (15NSd. EG. 1) 60 TO 32
2553
2550
               ** ITE (3,1134) $1,52,53,61,62,7M,5K,51GD,660,0643
0225
               CONTINUE
          50
0555
            32 WRITE (3,5000)
         0
         C
                 PLUT CEVIATURIC STRESS-STRAIN CLAVE ON
         C
                 X-Y RECONDER
               ## ITE (4, 1143)
CEET
2230
          1143 FURNAT( * X-Y PLUT X+CH+U, Y-CH#1, YES OR NO ")
               READ(4,2002) PLOT
1235
2232
               Y# ! #Y
               IF (FLOT . NE . Y) CALL EXIT
. . 35
:234
               CALL XYREC
0235
               STOP
2630
               END
```

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